A higher-order “figure-8” sensor and an isotropic sensor—For azimuth-elevation bivariate direction finding

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A “p-u probe” (also known as a “p-v probe”) comprises one pressure-sensor (which is isotropic) and one uni-axial particle-velocity sensor (which has a “figure-8” bi-directional spatial directivity). This p-u probe may be generalized, by allowing the figure-8 bi-directional sensor to have a higher order of directivity. This higher-order p-u probe has not previously been investigated anywhere in the open literature (to the best knowledge of the present authors). For such a sensing system, this paper is first (1) to develop closed-form eigen-based signal-processing algorithms for azimuth-elevation direction finding; (2) to analytically derive the associated Cramér-Rao lower bounds (CRB), which are expressed explicitly in terms of the two constituent sensors’ spatial geometry and in terms of the figure-8 sensor’s directivity order; (3) to verify (via Monte Carlo simulations) the proposed direction-of-arrival estimators’ efficacy and closeness to the respective CRB. Here, the higher-order p-u probe’s two constituent sensors may be spatially displaced.

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I. INTRODUCTION

The “p-u probe” is an acoustical sensing system popular for air acoustics and underwater acoustics. The p-u probe consists of (a) an isotropic pressure-sensor, which is isotropic in its gain-response, (b) a uni-axial particle-velocity sensor, whose gain-response is bi-directional, like a “figure-8,” i.e., \( \cos(\gamma) \) in mathematical form, where \( \gamma \) represents the angle from the directional sensor’s axis. Such a uni-axial particle-velocity sensor measures one Cartesian component of the acoustical wavefield’s particle velocity vector.

For a listing of the key literature on p-u probes, please refer to Ref. 1.

This figure-8 directivity could be sharpened through the use of a higher-order sensor, to give a \( \cos^k(\gamma) \) directivity (see Sec. 8.5. of Ref. 2 and Sec. 5 of Chap. 2 in Ref. 3):

(i) If a \( k \)th-order figure-8 sensor is oriented along the \( x \) axis, its gain response equals \( \left[ \cos(\theta) \cos(\phi) \right]^k \). Here, \( \theta \in [0, \pi] \) represents the polar angle (also known as the azimuth angle), and \( \phi \in [0, 2\pi] \) denotes the azimuth angle measured from the positive \( x \) axis.

(ii) If oriented along the \( y \)-axis, the gain response becomes \( \left[ \sin(\theta) \sin(\phi) \right]^k \).

(iii) If oriented along the \( z \)-axis, the gain response becomes \( \cos^k(\theta) \).

Please see Ref. 4 for a brief discussion of higher-order figure-8 sensors. (The particle-velocity sensor has an order of \( k = 1 \), whereas the isotropic pressure-sensor has an order of \( k = 0 \).) This paper generalizes the customary p-u probe, by allowing the figure-8 sensor to have any arbitrarily higher (integer) order \( k \) of directivity.

Second-order figure-8 bi-directional acoustic sensors have been implemented in hardware in Refs. 5–16. Third-order figure-8 bi-directional acoustic sensors have been implemented in hardware in Refs. 10, 14, and 17. Fifth-order figure-8 bi-directional acoustic sensors have been implemented in hardware in Ref. 18. Other higher-order figure-8 bi-directional acoustic sensors have been implemented in hardware in Refs. 10–14. These hardware implementations of second-order or higher-order p-u probes, dating from 1942 to 2008, show that second-order/higher-order p-u probes are established yet current sensing systems with continuing practical relevance.

Specifically, suppose that the isotropic sensor (i.e., a pressure-sensor) is placed at the Cartesian origin, and suppose that the figure-8 sensor lies on one of the three Cartesian axes and is also oriented in parallel to one of the Cartesian axes. Then, there would be nine distinct combinations of the figure-8 sensor’s location and axial orientation. Please see Fig. 1. Figure 1(a), for example, corresponds to a pressure sensor at the Cartesian origin, with a figure-8 directional sensor at the Cartesian position of \((\Delta_x, 0, 0)\) but orienting along the \( x \) axis. Figure 1(b), in contrast, has the figure-8 directional sensor at...
the Cartesian position of \((0, \Delta_y, 0)\) but still orienting along the \(x\) axis.

For each of these nine configurations at any specific sensor-order \(k\): Section II will define its array manifold. Section III will derive a new closed-form estimator of the incident source’s azimuth-elevation bivariate direction-of-arrival (or, for three of the nine array configurations, will explain why such an estimator is mathematically impossible). Section IV will analytically derive the corresponding Cramér-Rao lower bound, in a simple mathematical form that is explicitly in terms of the array geometry and explicitly in terms of sensor order \(k\). Section V will then present Monte Carlo simulations of the proposed estimator, showing its closeness to the derived Cramér-Rao lower bound. Last, Sec. VI will conclude the entire paper.

![FIG. 1. (Color online) The higher-order p-u probe’s nine configurations under investigation. The shaded configurations cannot facilitate bivariate azimuth-elevation direction-of-arrival estimation. Please see Sec. III. (a) \(V_x\) at \((\Delta_x, 0, 0)\), (b) \(V_x\) at \((0, \Delta_y, 0)\), (c) \(V_x\) at \((0, 0, \Delta_z)\), (d) \(V_y\) at \((0, \Delta_y, 0)\), (e) \(V_y\) at \((0, 0, \Delta_z)\), (f) \(V_y\) at \((0, 0, \Delta_z)\), (g) \(V_z\) at \((\Delta_x, 0, 0)\), (h) \(V_z\) at \((0, \Delta_y, 0)\), (i) \(V_z\) at \((0, 0, \Delta_z)\).](image)

### TABLE I. The array manifold for various configurations of \(k\)-th order p-u probe. [Please see Eq. (1).]

<table>
<thead>
<tr>
<th>Configuration</th>
<th>Equation</th>
</tr>
</thead>
<tbody>
<tr>
<td>Configuration (a):</td>
<td>(a^{(i)}_{P,V_x}(\theta, \phi) = \frac{1}{u^i} e^{i2\pi(\Delta_x/k)})</td>
</tr>
<tr>
<td>Configuration (b):</td>
<td>(a^{(i)}_{P,V_x}(\theta, \phi) = \frac{1}{v^i} e^{i2\pi(\Delta_y/k)})</td>
</tr>
<tr>
<td>Configuration (c):</td>
<td>(a^{(i)}_{P,V_x}(\theta, \phi) = \frac{1}{w^i} e^{i2\pi(\Delta_z/k)})</td>
</tr>
<tr>
<td>Configuration (d):</td>
<td>(a^{(i)}_{P,V_y}(\theta, \phi) = \frac{1}{u^i} e^{i2\pi(\Delta_x/k)})</td>
</tr>
<tr>
<td>Configuration (e):</td>
<td>(a^{(i)}_{P,V_y}(\theta, \phi) = \frac{1}{v^i} e^{i2\pi(\Delta_y/k)})</td>
</tr>
<tr>
<td>Configuration (f):</td>
<td>(a^{(i)}_{P,V_y}(\theta, \phi) = \frac{1}{w^i} e^{i2\pi(\Delta_z/k)})</td>
</tr>
<tr>
<td>Configuration (g):</td>
<td>(a^{(i)}_{P,V_z}(\theta, \phi) = \frac{1}{u^i} e^{i2\pi(\Delta_x/k)})</td>
</tr>
<tr>
<td>Configuration (h):</td>
<td>(a^{(i)}_{P,V_z}(\theta, \phi) = \frac{1}{v^i} e^{i2\pi(\Delta_y/k)})</td>
</tr>
<tr>
<td>Configuration (i):</td>
<td>(a^{(i)}_{P,V_z}(\theta, \phi) = \frac{1}{w^i} e^{i2\pi(\Delta_z/k)})</td>
</tr>
</tbody>
</table>
TABLE II. Closed-form estimates ($\hat{\theta}$, $\hat{\phi}$) of the azimuth-elevation bivariate direction-of-arrival, for the various configurations.

<table>
<thead>
<tr>
<th>#</th>
<th>“figure-8” sensor’s location</th>
<th>“figure-8” sensor’s orientation</th>
<th>($\hat{\theta}$, $\hat{\phi}$)</th>
</tr>
</thead>
<tbody>
<tr>
<td>(a)</td>
<td>$(\Delta_x, 0, 0)$</td>
<td>$x$-axis</td>
<td>$\hat{\theta}$, $\hat{\phi}$ unobtainable.</td>
</tr>
<tr>
<td></td>
<td></td>
<td></td>
<td>$\hat{\theta} = \begin{cases} \sin^{-1} \left( \text{sgn}(u) \sec(\phi) \left</td>
</tr>
<tr>
<td></td>
<td></td>
<td></td>
<td>$\hat{\phi} = \begin{cases} \frac{\pi}{2} \left[ \text{sgn}(u) - \text{sgn}(\nu) \right] \ +\tan^{-1} \left( \frac{1}{2\pi \Delta_x} \left( \text{sgn}(u) \left</td>
</tr>
<tr>
<td>(b)</td>
<td>$(0, \Delta_y, 0)$</td>
<td>$x$-axis</td>
<td>$\hat{\theta} = \cos^{-1} \left( \frac{1}{2\pi \Delta_x} \right) \left( \text{sgn}(u) \left</td>
</tr>
<tr>
<td></td>
<td></td>
<td></td>
<td>$\hat{\phi} = \begin{cases} \cos^{-1} \left( \csc(\theta) \left</td>
</tr>
<tr>
<td>(c)</td>
<td>$(0, 0, \Delta_z)$</td>
<td>$x$-axis</td>
<td>$\hat{\theta} = \begin{cases} \sin^{-1} \left( \text{sgn}(\nu) \csc(\phi) \left</td>
</tr>
<tr>
<td></td>
<td></td>
<td></td>
<td>$\hat{\phi} = \begin{cases} \frac{\pi}{2} \left[ \text{sgn}(u) - \text{sgn}(\nu) \right] \ +\tan^{-1} \left( \frac{2\pi \Delta_z}{\lambda} \left( \text{sgn}(\nu) \left</td>
</tr>
<tr>
<td>(d)</td>
<td>$(\Delta_x, 0, 0)$</td>
<td>$y$-axis</td>
<td>$\hat{\theta} = \begin{cases} \sin^{-1} \left( \text{sgn}(\nu) \sec(\phi) \left</td>
</tr>
<tr>
<td></td>
<td></td>
<td></td>
<td>$\hat{\phi} = \begin{cases} \frac{\pi}{2} \left[ \text{sgn}(u) - \text{sgn}(\nu) \right] \ +\tan^{-1} \left( \frac{2\pi \Delta_x}{\lambda} \left( \text{sgn}(\nu) \left</td>
</tr>
<tr>
<td>(e)</td>
<td>$(0, \Delta_y, 0)$</td>
<td>$y$-axis</td>
<td>$\hat{\theta}$, $\hat{\phi}$ unobtainable.</td>
</tr>
<tr>
<td></td>
<td></td>
<td></td>
<td>$\hat{\theta} = \cos^{-1} \left( \frac{1}{2\pi \Delta_x} \left( \text{sgn}(\nu) \left</td>
</tr>
<tr>
<td></td>
<td></td>
<td></td>
<td>$\hat{\phi} = \begin{cases} \pi \left[ \text{sgn}(\nu) - \text{sgn}(\nu) \right] \ +\tan^{-1} \left( \frac{2\pi \Delta_y}{\lambda} \left( \text{sgn}(\nu) \left</td>
</tr>
</tbody>
</table>

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This paper thus presents the higher-order p-u probe as an alternative to existing sensing systems, like the “v-v probe,”\textsuperscript{19} or the tri-axial velocity sensor,\textsuperscript{20,21} or the four-component acoustic vector-sensor (AVS) consisting of a tri-axial velocity sensor and a pressure-sensor.\textsuperscript{22–24,33,34} The use of a pressure sensor (instead of additional figure-8 sensors) may simplify the hardware and any calibration.

II. THE ARRAY MANIFOLDS OF THE GENERALIZED P-U PROBE’S NINE CONFIGURATIONS UNDER INVESTIGATION

Any location/orientation configuration of Fig. 1 may have its $2 \times 1$ array manifold represented as

$$
a_{P,V}^{[c]}(\theta, \phi) = \begin{bmatrix} 1 \\
\eta_1 e^{2\pi(\Delta_x/\xi_2)} \end{bmatrix}
$$

The first entry refers to the isotropic sensor at the Cartesian origin, whereas the second entry corresponds to the figure-8 directional sensor placed away from the Cartesian origin. Here, the superscript $\varepsilon \in \{x, y, z\}$ identifies the Cartesian axis on which the figure-8 sensor lies, the subscript $\zeta \in \{x, y, z\}$ indicates the orientation of figure-8 sensor, and

$$
\eta_1 := \begin{cases} 
sin(\theta) \cos(\phi), & \text{if} \quad \zeta = x; \\
\sin(\theta) \sin(\phi), & \text{if} \quad \zeta = y; \\
\cos(\theta), & \text{if} \quad \zeta = z; \\
\end{cases}
$$

$$
\eta_2 := \begin{cases} 
sin(\theta) \cos(\phi), & \text{if} \quad \varepsilon = x; \\
\sin(\theta) \sin(\phi), & \text{if} \quad \varepsilon = y; \\
\cos(\theta), & \text{if} \quad \varepsilon = z; \\
\end{cases}
$$

The first entry’s magnitude equals unity, on account of the pressure-sensor’s isotropcity. It has no complex phase,
because of its location at the Cartesian origin, hence no spatial phase factor. The second entry’s magnitude of $\eta_i^1$ corresponds to the $l$th-order figure-8 gain pattern oriented along the $\tau$ Cartesian coordinate. The second entry’s complex phase $e^{2\pi i (\Delta_c / 2)}$ represents a spatial phase factor for the figure-8 sensor’s location of $\Delta_c \approx 0$ on the $c$ Cartesian coordinate. The sign of the real-valued scalar $\eta_i$ specifies the hemisphere from which the source impinges. For example, at $\zeta = z$, $\text{sgn}(\eta_1) = \text{sgn}(\cos(\theta)) > 0$ would mean that the source impinges from the upper hemisphere, whereas $\text{sgn}(\eta_1) = \text{sgn}(\cos(\theta)) < 0$ means the lower hemisphere. Here, $\text{sgn}(\cdot)$ refers to the sign of the real-valued scalar inside the parentheses.

Figure 1’s nine location/orientation configurations’ array manifolds are presented in Table I. These nine array manifolds are functionally inter-related:

\[ (b) \leftrightarrow (d): \mathbf{a}_{pV_i}^{(i)}(\theta, \phi) = \mathbf{a}_{pV_i}^{(y)} \left( \theta, \frac{\pi}{2} - \phi \right), \]

\[ (c) \leftrightarrow (f): \mathbf{a}_{pV_i}^{(i)}(\theta, \phi) = \mathbf{a}_{pV_i}^{(z)} \left( \theta, \frac{\pi}{2} - \phi \right), \]

\[ (g) \leftrightarrow (h): \mathbf{a}_{pV_i}^{(y)}(\theta, \phi) = \mathbf{a}_{pV_i}^{(x)} \left( \theta, \frac{\pi}{2} - \phi \right), \]

\[ (e) \leftrightarrow (a): \mathbf{a}_{pV_i}^{(y)}(\theta, \phi) = \mathbf{a}_{pV_i}^{(z)} \left( \theta, \frac{\pi}{2} - \phi \right), \]

\[ (i) \leftrightarrow (e): \mathbf{a}_{pV_i}^{(y)}(\theta, \phi) = \mathbf{a}_{pV_i}^{(x)} \left( \frac{\pi}{2} - \theta, 0 \right) \]

Last, if the locations of two component-sensors are interchanged, the resulting array manifold is obtainable from the old one by a complex conjugation, then a multiplication by $e^{2\pi i (\Delta_c / 2)}$.

### III. EIGEN-BASED CLOSED-FORM ESTIMATION OF THE AZIMUTH-ELEVATION DIRECTION-OF-ARRIVAL

Eigen-based direction-of-arrival estimation involves an intermediate algorithmic step, wherein the incident signal’s steering vector is estimated to within an unknown complex-valued scalar $c$. This unknown $c$ arises from the eigen-decomposition of the data-correlation matrix. (Suppose $\mathbf{e}$ is an eigenvector of the data-correlation matrix, then $c \mathbf{e}$ must also be a valid eigenvector $\forall c \neq 0$.) That is, available to subsequent algorithmic steps is a steering-vector estimate, $\mathbf{a}_{pV_i}^{(c)}$.

In the ideal case of no noise or an infinite number of time samples, this approximation would become equality.

Hence, the problem is how to estimate $\theta$ and $\phi$, given $\mathbf{a}_{pV_i}^{(c)}$, for each of the nine configurations in Fig. 1 and Table I, with $k$ being any natural number that is prior known.

The unknown scalar $c$ may be eliminated as follows, on account of Eq. (1):

\[ \frac{[\mathbf{a}_{pV_i}^{(c)}]}{[\mathbf{a}_{pV_i}^{(c)}]} = \eta_1^1 e^{2\pi i (\Delta_c / 2)}, \]

where $[\cdot]_i$ denotes the $i$th element of the vector inside the square brackets.

For any prior known $\Delta_c \in (0, \lambda / 4]$, Eq. (8) leads to

### TABLE III. Cramér-Rao bounds for the generalized p-u probe’s nine configurations. Here, $\Lambda := 2\pi(\Delta_c / \lambda)$.

<table>
<thead>
<tr>
<th>Configuration</th>
<th>$2M_p^2 CRB_{\theta}^{(x)}$</th>
<th>$2M_p^2 CRB_{\phi}^{(x)}$</th>
</tr>
</thead>
<tbody>
<tr>
<td>(a) $V_i$ at $(\Delta_c, 0, 0)$</td>
<td>$\infty$</td>
<td>$\infty$</td>
</tr>
<tr>
<td>(b) $V_i$ at $(0, \Delta_c, 0)$</td>
<td>$\frac{\cos^2(\phi) \sin^2(\theta) + 1}{\Delta^2 \cos^4(\phi) \sin^2(\theta) + k^2 \cos^2(\phi)}$</td>
<td>$\frac{\cos^2(\phi) \sin^2(\theta) + 1}{\Delta^2 \sin^2(\theta) \sin^2(\phi) + k^2}$</td>
</tr>
<tr>
<td>(c) $V_i$ at $(0, 0, \Delta_c)$</td>
<td>$\frac{1 + \cos^2(\phi) \sin^2(\theta)}{\Delta^2 \cos^2(\phi) \sin^2(\theta) + k^2 \cos^2(\phi)}$</td>
<td>$\frac{\cos^2(\phi) \sin^2(\theta) + 1}{\Delta^2 \sin^2(\theta) + k^2 \cos^2(\phi)}$</td>
</tr>
<tr>
<td>(d) $V_i$ at $(0, \Delta_c, 0)$</td>
<td>$\frac{\sin^2(\phi) \sin^2(\theta) + 1}{\Delta^2 \cos^2(\phi) \sin^2(\theta) + k^2 \cos^2(\phi)}$</td>
<td>$\frac{\sin^2(\phi) \sin^2(\theta) + 1}{\Delta^2 \sin^2(\theta) + k^2 \cos^2(\phi)}$</td>
</tr>
<tr>
<td>(e) $V_i$ at $(0, 0, \Delta_c)$</td>
<td>$\infty$</td>
<td>$\infty$</td>
</tr>
<tr>
<td>(f) $V_i$ at $(0, 0, \Delta_c)$</td>
<td>$\frac{1 + \cos^2(\phi) \sin^2(\theta)}{\Delta^2 \cos^2(\phi) \sin^2(\theta) + k^2 \cos^2(\phi)}$</td>
<td>$\frac{\cos^2(\phi) \sin^2(\theta) + 1}{\Delta^2 \sin^2(\theta) + k^2 \cos^2(\phi)}$</td>
</tr>
<tr>
<td>(g) $V_i$ at $(\Delta_c, 0, 0)$</td>
<td>$\frac{\cos^2(\phi) \sin^2(\theta) + 1}{\Delta^2 \sin^2(\theta) \sin^2(\phi) + k^2 \cos^2(\phi)}$</td>
<td>$\frac{\cos^2(\phi) \sin^2(\theta) + 1}{\Delta^2 \sin^2(\theta) \sin^2(\phi) + k^2 \cos^2(\phi)}$</td>
</tr>
<tr>
<td>(h) $V_i$ at $(\Delta_c, 0, 0)$</td>
<td>$\frac{\cos^2(\phi) \sin^2(\theta) + 1}{\Delta^2 \sin^2(\theta) \sin^2(\phi) + k^2 \cos^2(\phi)}$</td>
<td>$\frac{\cos^2(\phi) \sin^2(\theta) + 1}{\Delta^2 \sin^2(\theta) \sin^2(\phi) + k^2 \cos^2(\phi)}$</td>
</tr>
<tr>
<td>(i) $V_i$ at $(0, 0, \Delta_c)$</td>
<td>$\frac{\cos^2(\phi) \sin^2(\theta) + 1}{\Delta^2 \sin^2(\theta) \sin^2(\phi) + k^2 \cos^2(\phi)}$</td>
<td>$\frac{\cos^2(\phi) \sin^2(\theta) + 1}{\Delta^2 \sin^2(\theta) \sin^2(\phi) + k^2 \cos^2(\phi)}$</td>
</tr>
</tbody>
</table>
\( \hat{\eta}_1 = \left( \frac{\mathbb{A}_{P,V_c}^{(c)}}{\mathbb{A}_{P,V_c}^{(c)}} \right)^{1/k} \left( \frac{\mathbb{A}_{P,V_c}^{(c)}}{\mathbb{A}_{P,V_c}^{(c)}} \right) \),
\( \hat{\eta}_2 = \frac{\lambda}{2\pi\Delta_e} \left( \text{sgn}(\eta_1) \frac{\mathbb{A}_{P,V_c}^{(c)}}{\mathbb{A}_{P,V_c}^{(c)}} \right). \)

If \( \Delta_e > \lambda/2 \), the spatial phase factor \( e^{j(2\pi\Delta_e/\lambda)\eta_2} \) has no one-to-one mapping with \( \eta_2 \); hence, Eqs. (9) and (10) cannot uniquely estimate both \( \theta \) and \( \phi \). However, the extended-aperture methodology can be used to resolve the cases with \( \Delta_e > \lambda/2 \).

From the above \( \hat{\eta}_1 \) and \( \hat{\eta}_2 \), the closed-form estimates \( \hat{\theta} \) and \( \hat{\phi} \) are specified in Table II, for \( \epsilon \neq \zeta \).

There, due to \( \text{sgn}(\eta_1) \) and due to the cyclic ambiguities of inverse trigonometric functions inherent in \( \eta_1 \) and \( \eta_2 \), \( \hat{\theta} \) and \( \hat{\phi} \) can be unambiguous for only a tetarto-sphere (i.e., a quarter of a sphere). For configuration (b), the necessary prior knowledge is whether \( \theta \in [0, \pi/2) \) or \( \theta \in [\pi/2, \pi) \) (i.e., upper vs lower hemisphere) and whether \( u > 0 \) or \( u < 0 \) (i.e., front vs back hemisphere). For configuration (c), the necessary prior knowledge is whether \( \phi \in [0, \pi) \) or \( \phi \in [\pi, 2\pi) \) (i.e., right vs left hemisphere) and whether \( u > 0 \) or \( u < 0 \) (i.e., left vs right hemisphere). For configuration (d), the necessary prior knowledge is whether \( \phi \in [0, \pi/2) \) or \( \phi \in [\pi/2, \pi) \) (i.e., upper vs lower hemisphere) and whether \( u > 0 \) or \( u < 0 \) (i.e., front vs back hemisphere). For configuration (f), the necessary prior knowledge is whether \( \phi \in [-\pi/2, \pi/2) \) or \( \phi \in [\pi/2, 3\pi/2) \) (i.e., front vs back hemisphere) and whether \( u > 0 \) or \( u < 0 \) (i.e., right vs left hemisphere). For configuration (f), the necessary prior knowledge is whether \( \phi \in [-\pi/2, \pi/2) \) or \( \phi \in [\pi/2, 3\pi/2) \).
hemisphere). For configuration (g), the necessary prior knowledge is whether $\phi \in [0, \pi) \text{ or } \phi \in [\pi, 2\pi)$ (i.e., right vs left hemisphere) and whether $w > 0 \text{ or } w < 0$ (i.e., upper vs lower hemisphere). For configuration (h), the necessary prior knowledge is whether $\phi \in [-\pi/2, \pi/2)$ or $\phi \in [\pi/2, 3\pi/2)$ (i.e., front vs back hemisphere) and whether $w > 0 \text{ or } w < 0$ (i.e., upper vs lower hemisphere).

At $\theta = 0$, $\pi$, the impinging signal has none of its energy projected onto the $x$-$y$ plane; hence, $\phi$ would be impossible by any estimator.

If $\eta_1 = \eta_2$, or equivalently if $\varepsilon = \zeta$ [as in configurations (a) and (e) and (i)], the system of equations in Eqs. (9) and (10) would be indeterminable, because the right sides of Eqs. (9) and (10) would be equal, thereby offering only one constraint for two unknowns.

If the two sensors’ have their locations switched: The array manifold’s multiplicative factor $(\epsilon^{2\pi(i/\lambda)\eta_2})$ will be absorbed into the eigen-composition $c$, hence poses no change to the estimation formulas there. The array manifold’s complex conjugation would result simply in a sign change at the appropriate places of each estimator.

IV. THE CRAMÉR-RAO BOUND FOR THE KTH-ORDER P-U PROBE IN VARIOUS CONFIGURATIONS

The Cramér-Rao bound lower-bounds the error variance obtainable from any unbiased estimator, given the statistical model that connects the observed data to the unknown parameter being estimated. To focus on the directivity order $k$ and on the spatial configuration, the following analysis will
use a simple statistical model for the incident signal and for the corrupting noise. This analysis could be readily extended to more complicated signal/noise scenarios.

Model the incident signal as a pure-tone

\[ s(t) = \sqrt{P_s} e^{i(x t + \phi)}, \]  

(11)

where \( P_s \) denotes the signal power and \( \phi \) the initial phase, both deterministic but allowed to be unknown. Let there be additive noise, modeled as Gaussian, zero-mean, statistically uncorrelated over time and across the two component-sensors, with an unknown power of \( P_n \).

At the \( m \)th time instant of \( t = mT_s \) (where \( T_s \) denotes the time-sampling period), the p-u probe provides a \( 2 \times 1 \) data vector of

\[ \tilde{z}(mT_s) = a s(mT_s) + \tilde{n}(mT_s), \quad \forall m = 1, 2, ..., M. \]  

(12)

It is assumed that \( \omega, \Delta, \) and \( \lambda \) are previously known.

This statistical data model has five real-valued scalar unknowns: \( \theta, \phi, P_s, P_n, \phi \). Hence, the resulting Fisher

FIG. 6. (Color online) Configuration (g): CRB of \( \theta \) [\( \text{CRB}^{(g)}(\theta, \phi) \)] for various orders \( k \) of figure-8 sensor. (i) \( k = 1 \), (ii) \( k = 2 \), (iii) \( k = 3 \).

FIG. 7. (Color online) Configuration (g): CRB of \( \phi \) [\( \text{CRB}^{(g)}(\theta, \phi) \)] for various orders \( k \) of figure-8 sensor. (i) \( k = 1 \), (ii) \( k = 2 \), (iii) \( k = 3 \).

FIG. 8. (Color online) Symmetries in the figure-8 sensor’s gain response.
information matrix (FIM) is $5 \times 5$ in size. The corresponding Cramér-Rao bounds are derived (using Sec. 8.2.3.1 of Ref. 31) and stated in Table III. Therein, for example, the superscript in $\text{CRB} \left( \theta, \phi \right)$ refers to configuration (a), whereas the subscript $h$ identifies the to-be-estimated parameter whose Cramér-Rao bound is symbolized.

Figures 2–7 plot the Cramér-Rao bounds at various values of the sensor-order $k$, for the six configurations that allow bivariate azimuth-elevation direction-of-arrival estimation. The remaining three configurations, shaded in Fig. 1 and Table I, would not allow such bivariate direction finding, for reasons already explained at the end of Sec. III.

A. Inter-relationships among various configurations’ Cramér-Rao bounds

Because the various configurations’ array manifolds are functionally inter-related as in Eqs. (2)–(5), their corresponding Cramér-Rao bounds are also correspondingly inter-related as follows:

\[(b) \leftrightarrow (d) : \text{CRB}^{(b)}(\theta, \phi) = \text{CRB}^{(d)} \left( \theta, \frac{\pi}{2} - \phi \right), \quad (13)\]

\[(c) \leftrightarrow (f) : \text{CRB}^{(c)}(\theta, \phi) = \text{CRB}^{(f)} \left( \theta, \frac{\pi}{2} - \phi \right), \quad (14)\]

\[(g) \leftrightarrow (h) : \text{CRB}^{(g)}(\theta, \phi) = \text{CRB}^{(h)} \left( \theta, \frac{\pi}{2} - \phi \right), \quad (15)\]

\[(e) \leftrightarrow (a) : \text{CRB}^{(e)}(\theta, \phi) = \text{CRB}^{(a)} \left( \theta, \frac{\pi}{2} - \phi \right), \quad (16)\]

with the subscript $\times \in \{\theta, \phi\}$.

B. Symmetries in each Cramér-Rao bound

Each Cramér-Rao bound is

(i) symmetric with respect to $\theta = 90^\circ$ over $\theta \in \left[0^\circ, 180^\circ\right]$,

(ii) symmetric with respect to $\phi = 0^\circ$ over $\phi \in \left[-90^\circ, 90^\circ\right]$,

(iii) symmetric with respect to $\phi = 90^\circ$ over $\phi \in \left[0^\circ, 180^\circ\right]$,

(iv) symmetric with respect to $\phi = 180^\circ$ over $\phi \in \left[90^\circ, 270^\circ\right]$,

(v) symmetric with respect to $\phi = 270^\circ$ over $\phi \in \left[180^\circ, 360^\circ\right]$.

These symmetries arise mathematically in every Cramér-Rao bound expression due to the even powers to which the trigonometric functions are raised. Physically speaking, these symmetries exist because the figure-8 sensor’s gain response (a) has a longitudinal cross-section that is 360° rotationally invariant with regard to the figure-8 sensor’s axis, and (b) has two lobes that are left/right symmetric to each other. Please see Fig. 8.
FIG. 11. (Color online) Cumulative histograms to compare the six configurations with the figure-8 sensor at order $k = 3$, vs the customary four-component acoustic vector-sensor (AVS). (i) For $\theta$, (ii) For $\phi$.

FIG. 12. (Color online) Configuration (b): The proposed $\theta$'s RMSE vs $\sqrt{\text{CRB}}(\theta, \phi)$, at various sensor-orders $k$. (i) $k = 1$, (ii) $k = 2$, (iii) $k = 3$.

FIG. 13. (Color online) Configuration (b): The proposed $\phi$'s RMSE vs $\sqrt{\text{CRB}}(\theta, \phi)$, at various sensor-orders $k$. (i) $k = 1$, (ii) $k = 2$, (iii) $k = 3$. 

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C. To compare across the various configurations in Fig. 1 and Table I

To compare across various configurations by their direction-finding precision: Figs. 9–11 plot the Cramér-Rao bound’s cumulative histogram, which reveals the percentage of all possible incident direction-of-arrival from the two-dimensional support region of \( \{ \theta \in [0, \pi] \} \cup \{ \phi \in [0, 2\pi] \} \), at which the corresponding configuration can estimate better than or equal to the precision specified by the abscissa. Figure 9 is for sensor-order \( k = 1 \), Fig. 10 for \( k = 2 \), and Fig. 11 for \( k = 3 \). Each figure compares all six configurations (i.e., all those that allow bivariate direction finding); nonetheless, the identities in Eqs. (13)–(15) imply only three distinct cumulative histogram curves for each of Figs. 9–11. These cumulative histograms are computed here using spatially uniform sampling over the unit-sphere’s surface.32 The number of spatial samples equals \( 1.5 \times 10^6 \) over the spherical surface. Recall that the Cramér-Rao bound represents the hypothetically best precision obtainable in estimating the direction-of-arrival: the smaller the Cramér-Rao bound the better, hence the higher the cumulative histogram the better.

Some qualitative observations on CRB \( \gamma (\theta, \phi) \):

1. Configuration (b)’s CRB\( \gamma (\theta, \phi) \) and configuration (d)’s CRB\( \gamma (\theta, \phi) \) are worse than the other configurations’ CRB\( \gamma (\theta, \phi) \). This is intuitively reasonable, because these two configurations:
   (i) orient the figure-8 directional sensor to yield no \( \theta \)-directivity, and
   (ii) space the two component-sensors to yield no vertical aperture.
(2) Configurations (g)’s and (h)’s cumulative histogram is crossed over by that of configurations (c) and (f), as the abscissa gets sufficiently large. Moreover, this cross-over abscissa drops, as the directivity order \( k \) increases. The explanation is as follows: Configurations (c) and (f) provide an inter-sensor spatial aperture along the vertical, but no vertical directivity exists in its figure-8 sensor. In contrast, configurations (g) and (h) offer vertical directivity but no vertical aperture. Figures 9–11 indicate that a \( \lambda/2 \) vertical aperture is more important than the vertical directivity, to attain a very low Cramér-Rao bound for selected \((\theta, \phi)\) sectors. A sufficiently high \( k \) could dominate any lack of aperture.

Some qualitative observations on CRB_\( \phi \) (\(\theta, \phi\)):

(3) For CRB_\( \phi \) (\(\theta, \phi\)), configurations (b) and (d) are better than the other configurations. This is also intuitively reasonable, because these two configurations

(i) orient the figure-8 directional sensor to maximize \( \phi \)-directivity, as well as

(ii) space the two component-sensors to give maximum horizontal inter-sensor aperture.

(4) Configurations (g) and (h)’s cumulative histogram crosses over that of configurations (c) and (f), as the abscissa gets sufficiently large. Moreover, this cross-over abscissa drops, as the directivity order \( k \) increases. The explanation is exactly same as that under point (2) above.

(5) The Cramér-Rao bounds would be unaffected by any switch of the two component-sensors’ locations.
The four-component acoustic vector-sensor (AVS), comprising a tri-axial velocity-sensor and a collocating pressure-sensor, offers better Cramér-Rao bounds, than any of the high-order p-u probes. This is unsurprising, because the acoustic vector-sensor has more component-sensors.

V. MONTE CARLO SIMULATIONS OF THE ESTIMATORS PROPOSED IN SEC. III

This section presents Monte Carlo simulations of the estimator proposed in Sec. III for configuration (b), as an illustrative example. These simulations show the estimator’s efficacy, with an estimation error variance very close to the Cramér-Rao bound derived in Sec. IV.

The statistical data model of Sec. IV is retained here. Moreover, $\Delta_{c}/\lambda = 1/2$, $\omega = 1885$ radians/s, and $\varphi = 0$.

FIG. 18. (Color online) Configuration (c): The proposed $\hat{\theta}$’s RMSE and $\sqrt{\text{CRB}}(\hat{\theta}, \phi)$ vs sensor’s separation $\Delta_{c}/\lambda$, at various sensor-orders $k$. (i) $k = 1$, (ii) $k = 2$, (iii) $k = 3$.

FIG. 19. (Color online) Configuration (c): The proposed $\hat{\phi}$’s RMSE and $\sqrt{\text{CRB}}(\hat{\theta}, \phi)$ vs sensor’s separation $\Delta_{c}/\lambda$, at various sensor-orders $k$. (i) $k = 1$, (ii) $k = 2$, (iv) $k = 3$.

Figures 12–15 compare configuration (b)’s estimators in the second row of Table II, against the corresponding Cramér-Rao bounds. These figures unanimously verify the proposed estimator’s efficacy and closeness to the Cramér-Rao bounds. Figures 16–19 do the same for configuration (c)’s estimators in the third row of Table II, against the corresponding Cramér-Rao bounds in the third row of Table III.

Figures 12 and 13 and 16 and 17 show that the Cramér-Rao bounds decrease (i.e., improve) as the SNR increases for $k = 1, 2, 3$, as would be expected. Figures 14 and 15 and 18 and 19 show that the Cramér-Rao bounds also decrease as the inter-sensor spacing $\Delta_{c}$ increases (and thus the array aperture is enlarged) for $k = 1, 2, 3$, as also would be expected.

Last:

(6) The four-component acoustic vector-sensor (AVS), comprising a tri-axial velocity-sensor and a collocating pressure-sensor, offers better Cramér-Rao bounds, than any of the high-order p-u probes. This is unsurprising, because the acoustic vector-sensor has more component-sensors.
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25Suppose a signal $s(t)$ reaches the u-p probe, but becomes corrupted additively by the p-u probe’s thermal noise-vector $n(t)$. The p-u probe’s measurement then equals to $2 \times 1$ data vector of $x(t) = x(t) + n(t)$ at the sample $t$. From $M$ such time samples, a data correlation matrix of $C = \sum_{t=1}^{M} x(t)x(t)^H$ is formed, where the superscript $H$ symbolizes the Hermitian operator. Suppose further that $(s(t))$ and $(n(t))$ are each temporally stationary and not cross-correlated between them. Then, $C \approx C = MP_{uv} + MP_{u}I$, where $P_{uv}$ denotes the power of the incident signal, a symbolizes the impinging source’s steering vector, $P_{u}$ refers to the thermal noise power at each component-sensor, and $I$ signifies a $2 \times 2$ identity matrix. This $2 \times 2$ matrix $C$ is Hermitian, and asymptotically approaches $C$ as $M \to \infty$. The asymptotic $C$ has a principal eigenvector equal to $c$, where $c$ is some complex-valued scalar that is algebraically independent of $a$.

26If $\Delta_0 = 0$ (i.e., if the two sensors are collocated at one point in space), $\Delta_2$ cannot be estimated in Eq. (10), even though $\Delta_1$ may still be estimated via Eq. (9).


